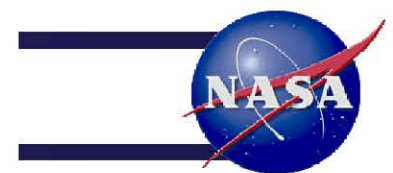




THE TRANSPORT OF MASS, ENERGY, AND ENTROPY IN CRYOGENIC SUPPORT STRUTS FOR ENGINEERING DESIGN



Thermal and Fluids Analysis Workshop

August 13-17, 2012, Pasadena, California

J.P. Elchert

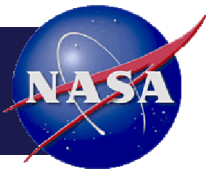
NASA Glenn Research Center

Thermal Systems Branch

August 8, 2012

1 Abstract

Engineers working to understand and reduce cryogenic boil-off must solve a variety of transport problems. An important class of nonlinear problems involves the thermal and mechanical design of cryogenic struts. These classic problems are scattered about the literature and typically require too many resources to obtain. So, to save time for practicing engineers, the author presents this essay. Herein, a variety of new, old, and revisited analytical and finite difference solutions of the thermal problem are covered in this essay, along with commentary on approach and assumptions. This includes a few thermal radiation and conduction combined mode solutions with a discussion on insulation, optimum emissivity, and geometrical phenomenon. Solutions to cooling and heat interception problems are also presented, including a discussion of the entropy generation. And the literature on the combined mechanical and thermal design of cryogenic support struts is reviewed with an introduction to the associated numerical methods.



The transport of mass, energy, and entropy in cryogenic support struts for engineering design

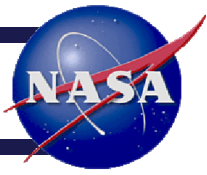
J.P. Elchert, NASA Glenn Research Center,
Thermal Systems Branch

Presented By
J.P. Elchert

Thermal & Fluids Analysis Workshop
TFAWS 2012
August 13-17, 2012
Jet Propulsion Laboratory
Pasadena, CA



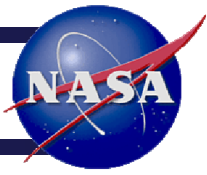
Table of Contents



- ✓ History
- ✓ State of the art
- ✓ Future

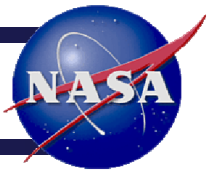


Royal Institution of Great Britain



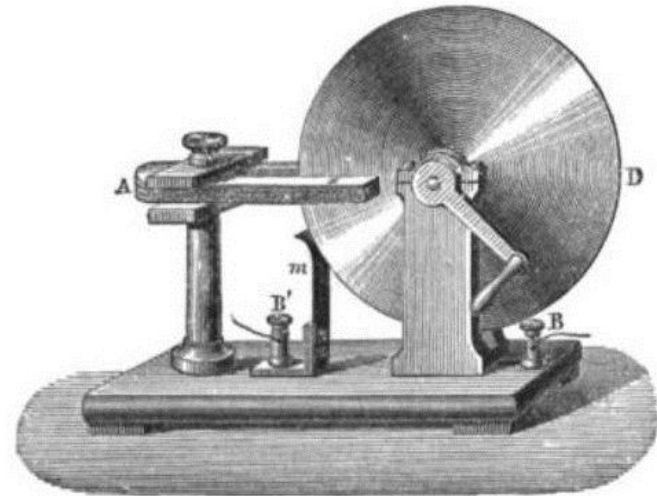


Standing on the shoulders of giants



$$\oint_{\partial\Sigma} \mathbf{E} \cdot d\boldsymbol{\ell} = - \iint_{\Sigma} \frac{\partial \mathbf{B}}{\partial t} \cdot d\mathbf{S}$$

1831, simple DC generator





Massachusetts's Institute of Technology



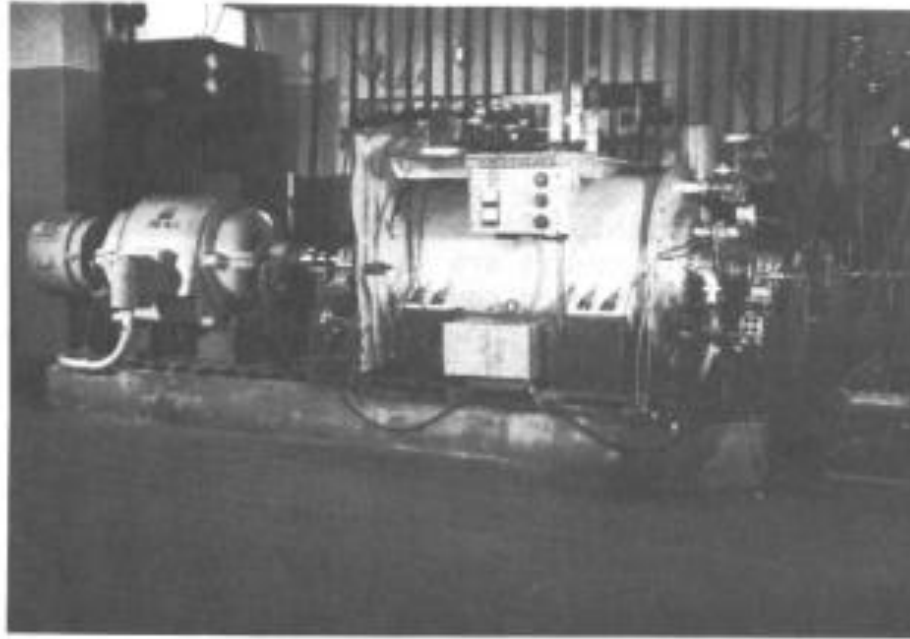
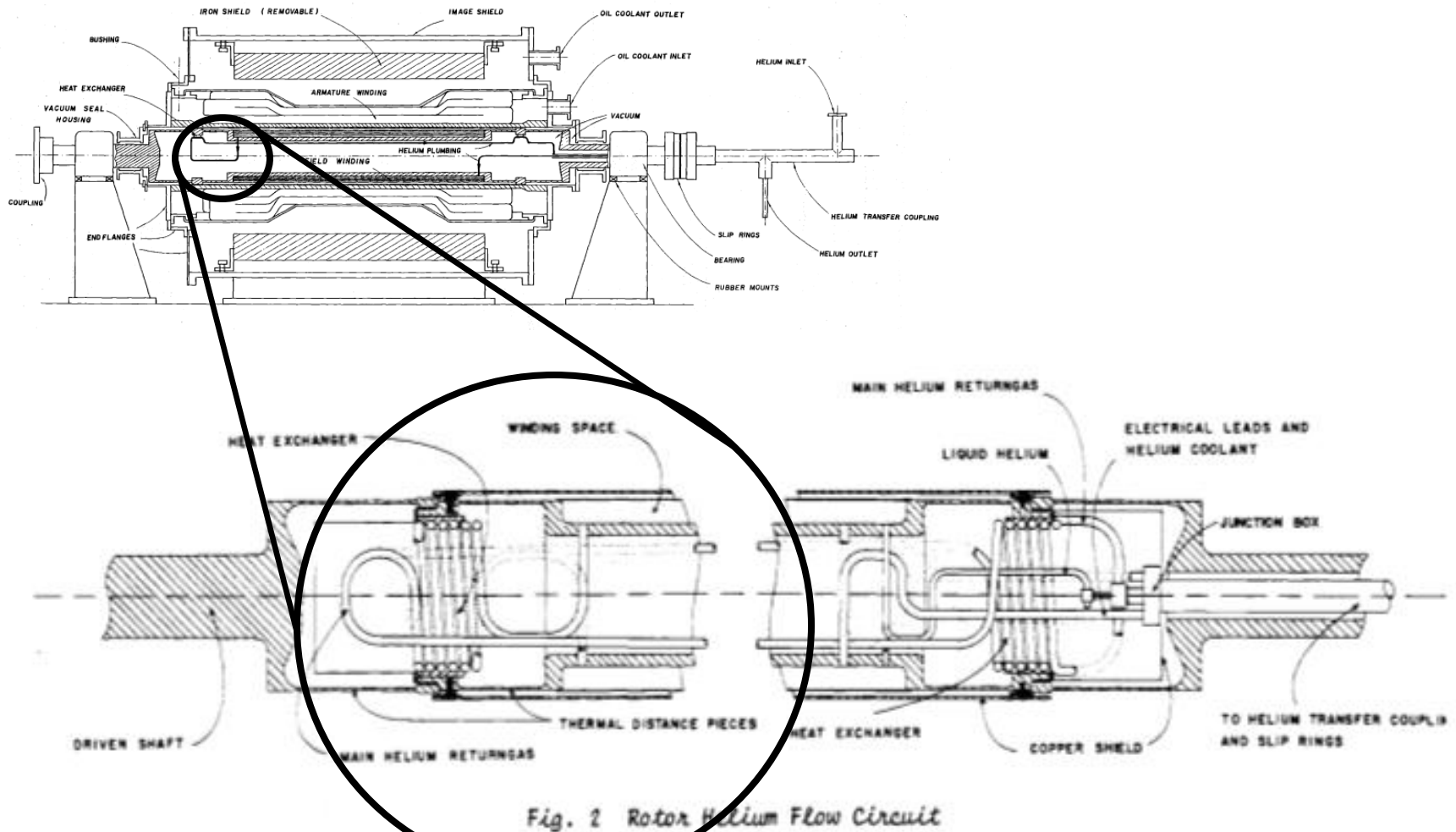


Fig. 9 View of 3-MVA Alternator and 150-HP Drive Motor

- In the 1970s, MIT Electric Power Research Institute did significant research into superconducting synchronous generators, including cryogenics engineering research



MIT's prototype synchronous generator



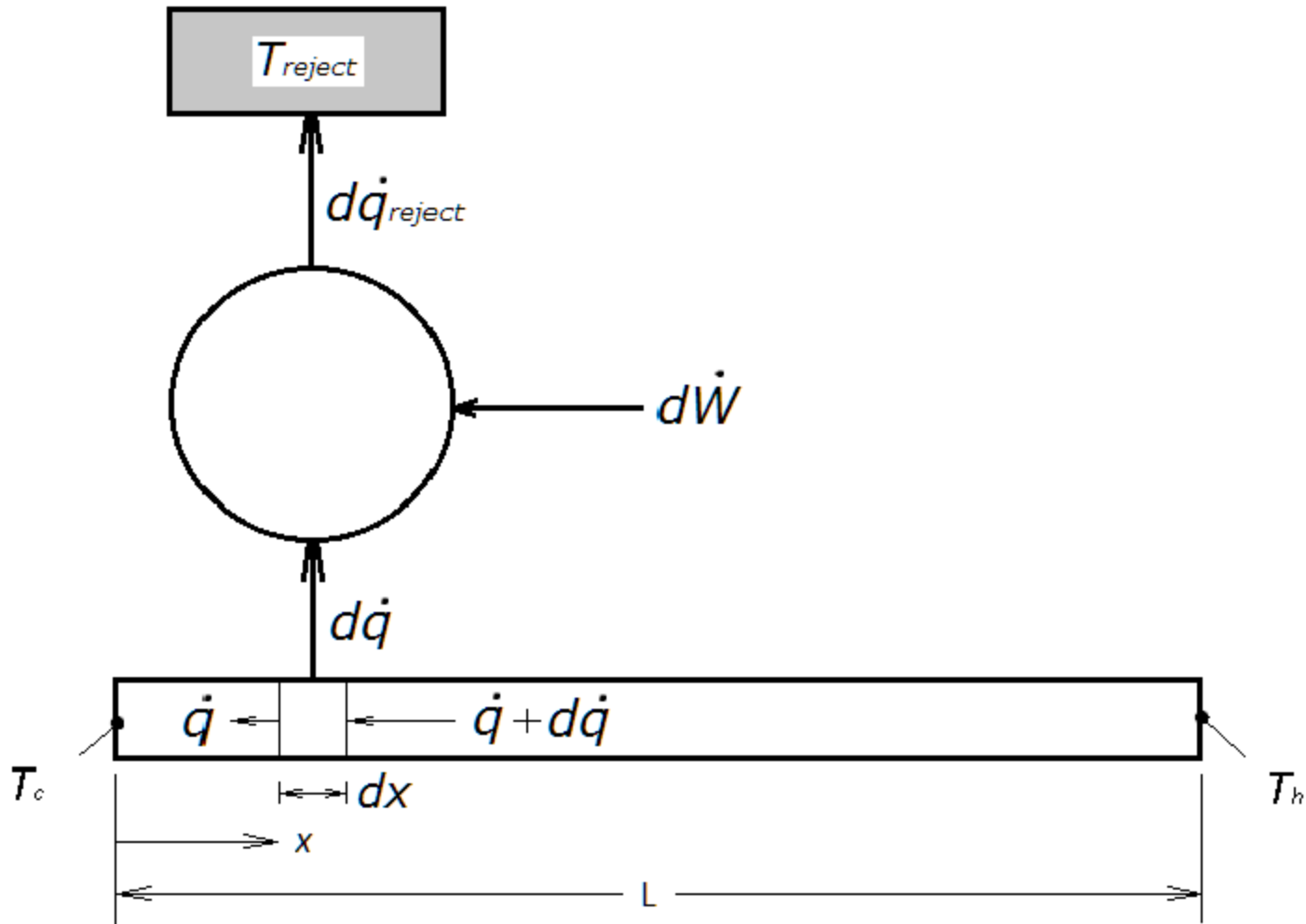
Dr. Adrian Bejan

J.A. Jones Professor, Duke Department of Mechanical Engineering and Materials Science



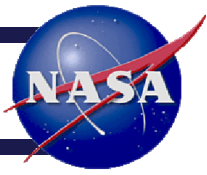


Adrian Bejan (1975)





Thermodynamic Minimum



$$\dot{S} = \frac{A}{L} \left(\int_{T_c}^{T_h} \frac{\sqrt{k}}{T} dT \right)^2 = \frac{kA}{L} \left[\ln \left(\frac{T_h}{T_c} \right) \right]^2$$



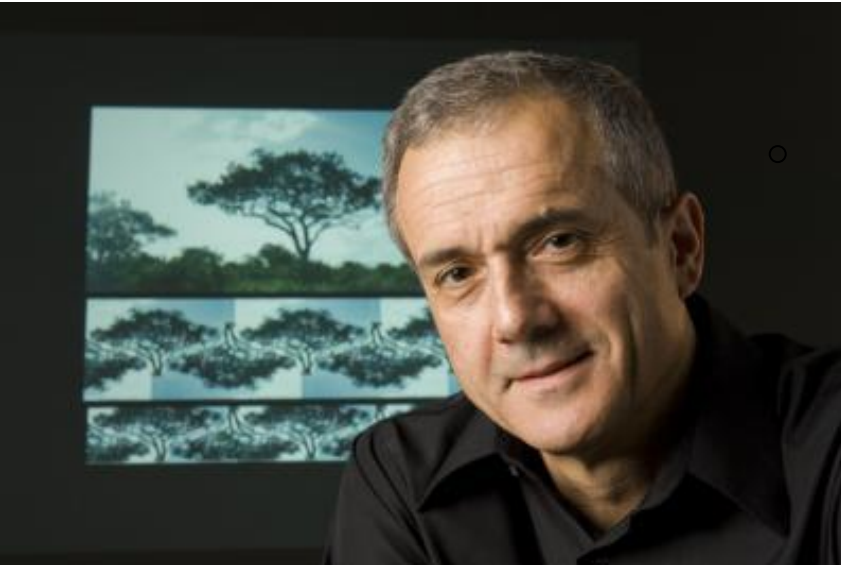
The most efficient cooling scheme generates the least entropy



$$\dot{q}_{optimum} = T\sqrt{k} \left(\frac{A}{L} \right) \left(\int_{T_c}^{T_h} \frac{\sqrt{k}}{T} dT \right)$$

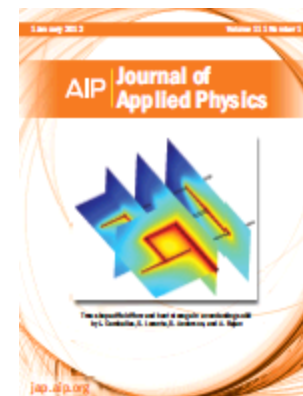
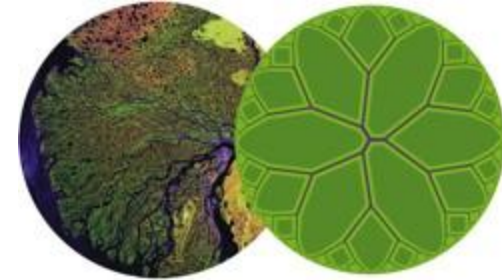
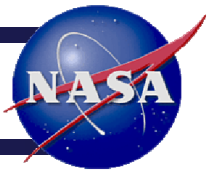


$$\dot{W} = \frac{A}{L} T_h \left(\int_{T_c}^{T_h} \frac{\sqrt{k}}{T} dT \right)^2 = T_h \frac{kA}{L} \left[\ln \left(\frac{T_h}{T_c} \right) \right]^2$$





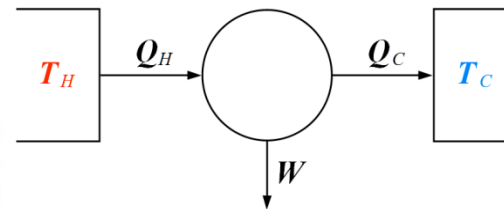
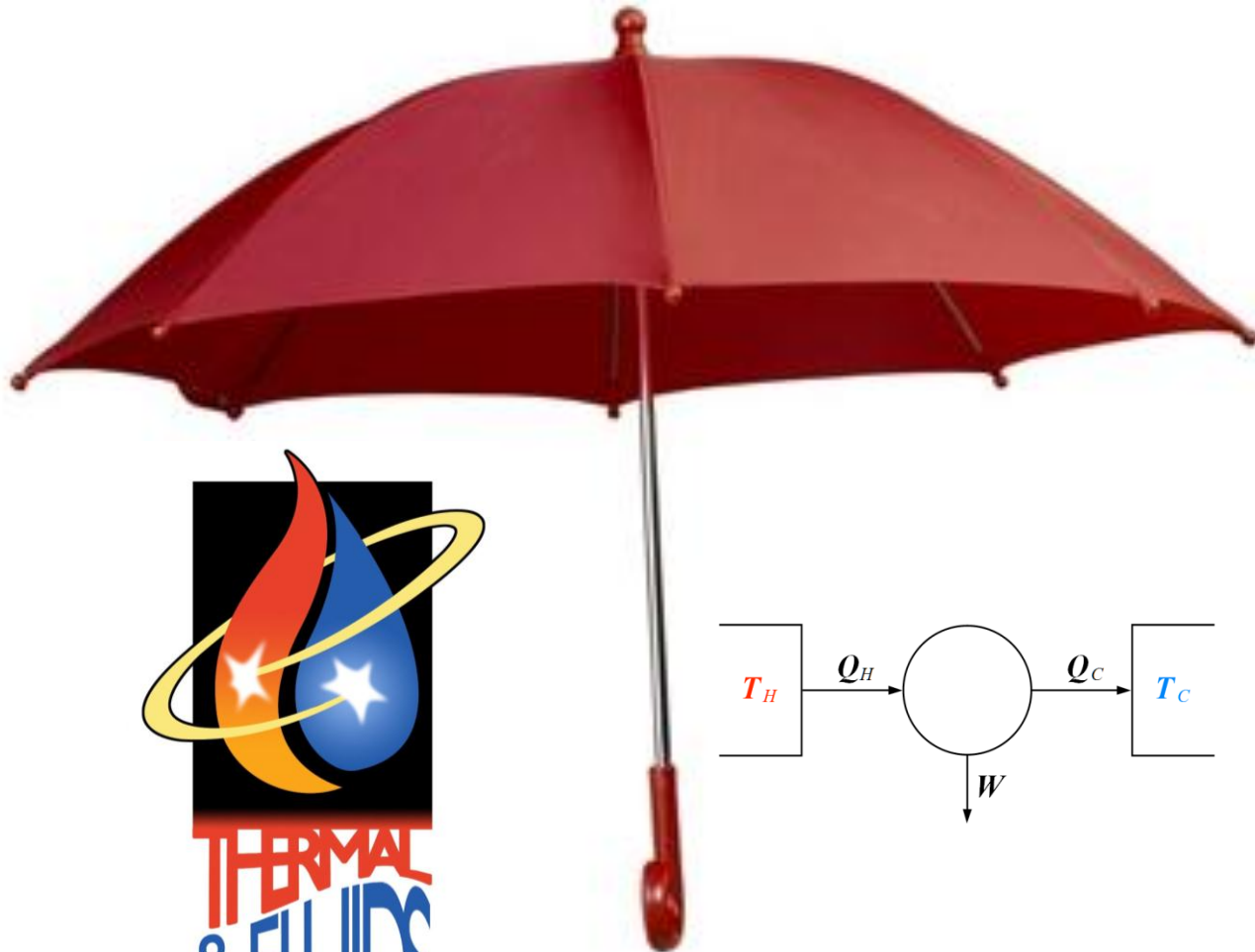
Constructal Law



Finite-size systems evolve progressively easier access to the imposed current

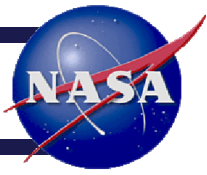


Unifies thermal/fluids/thermodynamics





Continuous cooling (solution by Tsao)



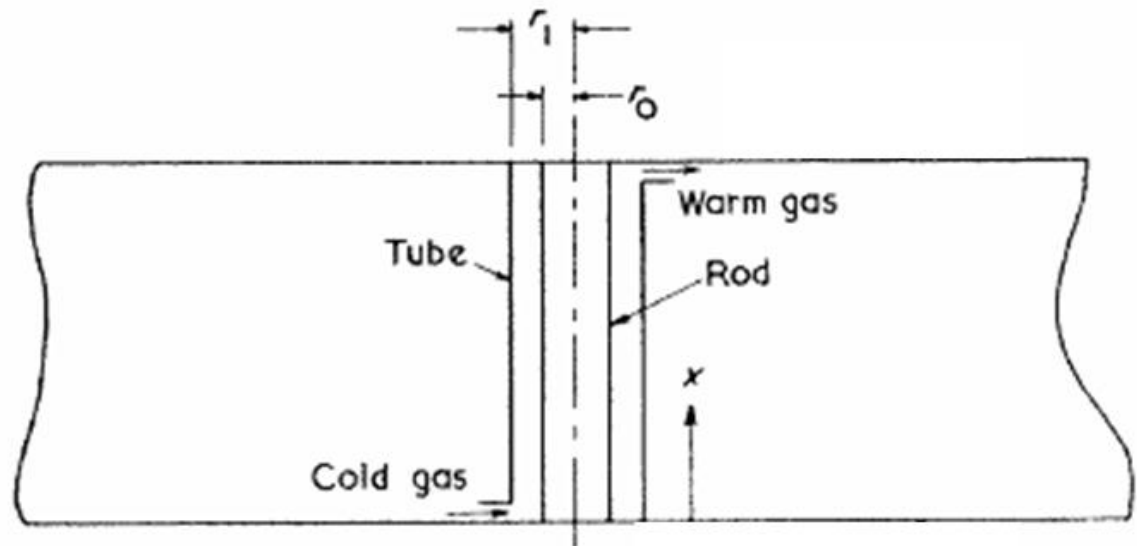
$$\frac{d}{dx} \left(A_c k(T) \frac{dT}{dx} \right) - Ph(T) [T(x) - T_m(x)] = 0$$

$$\dot{m} c_p \frac{dT_m}{dx} - Ph(T) [T(x) - T_m(x)] = 0$$

$$T(0) = T_L$$

$$T(L) = T_H$$

$$T_m(0) = T_o$$



$$\eta = x/L$$

$$\gamma^2 = 4\alpha^2 + \beta^2$$

$$\tau_f = \frac{(T_f - T_c)}{(T_h - T_c)}$$

$$\beta = \frac{hPL}{\dot{m}c_p}$$

$$\tau_s = \frac{(T_s - T_c)}{(T_h - T_c)} \quad \alpha = \left(\frac{L/(kA)}{1/(hPL)} \right)^{1/2}$$

Governing Equations

$$\frac{d^2\tau_s}{d\eta^2} = \alpha^2(\tau_s - \tau_f)$$

$$\frac{d\tau_f}{d\eta} = \beta(\tau_s - \tau_f)$$

Boundary conditions

$$T_s(x=0) = T_c \Rightarrow \tau_s(\eta=0) = \frac{T_s(x=0) - T_c}{T_h - T_c}$$

$$T_s(x=L) = T_h \Rightarrow \tau_s(\eta=1) = \frac{T_s(x=L) - T_c}{T_h - T_c}$$

$$T_f(x=0) = T_c \Rightarrow \tau_f(\eta=0) = \frac{T_f(x=0) - T_c}{T_h - T_c}$$

Solutions

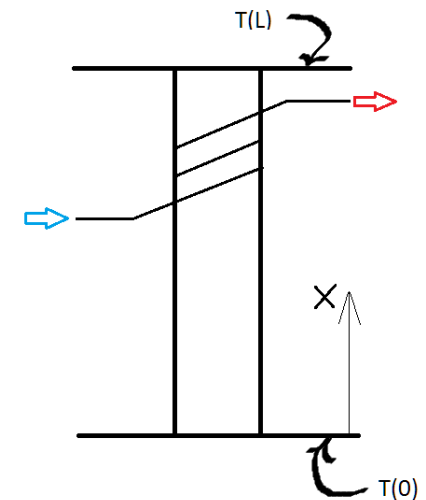
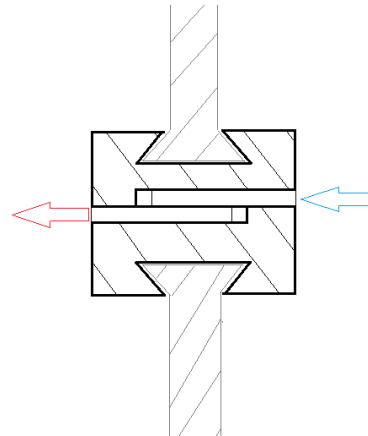
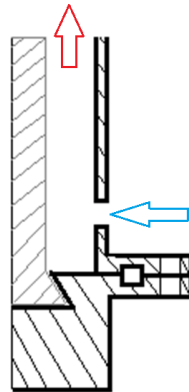
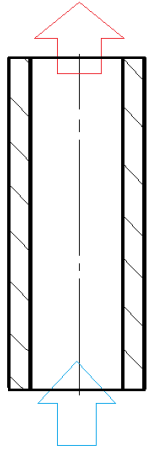
$$\tau_s = \frac{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}\eta} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}\eta}}{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}}}$$

$$\tau_f = 2\beta \frac{(\gamma + \beta)e^{\frac{\gamma - \beta}{2}\eta} - 2\gamma - (\gamma - \beta)e^{-\frac{\gamma + \beta}{2}\eta}}{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}}}$$

$$\frac{\dot{q}''}{k(T_h - T_c)/L} = \left. \frac{d\tau_s}{d\eta} \right|_{\eta \rightarrow 0} = \frac{\gamma(\gamma^2 - \beta^2)}{(\gamma + \beta)^2 e^{\frac{\gamma - \beta}{2}} - 4\gamma\beta + (\gamma - \beta)^2 e^{-\frac{\gamma + \beta}{2}}}$$



Various cooling schemes



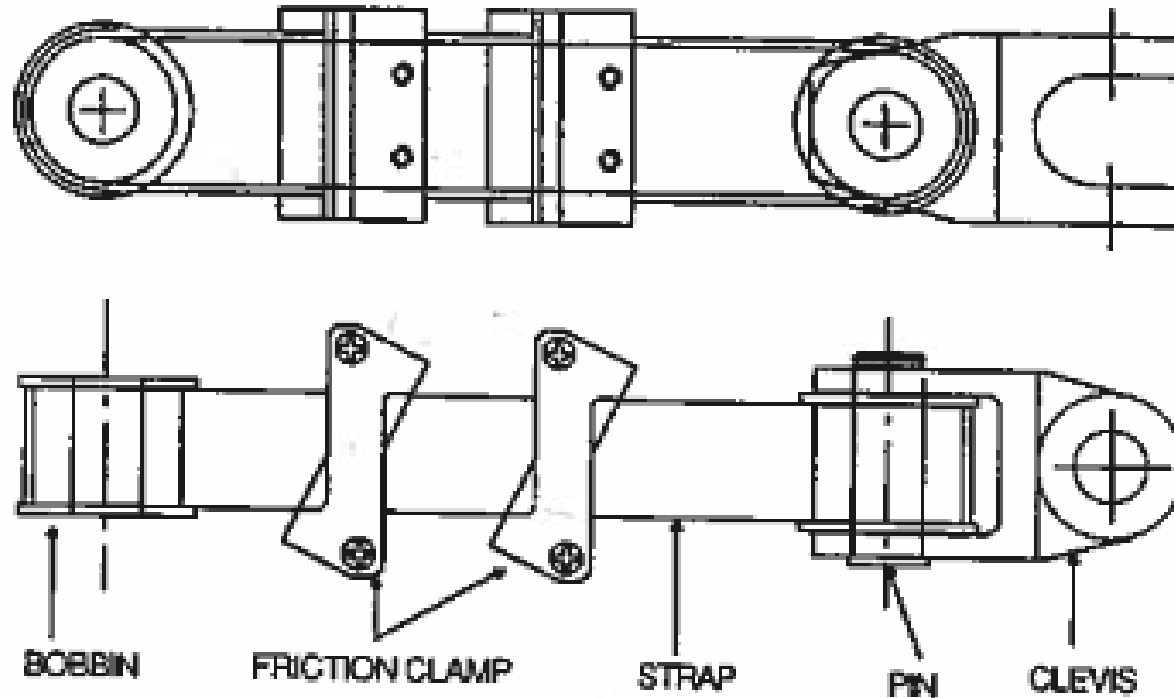
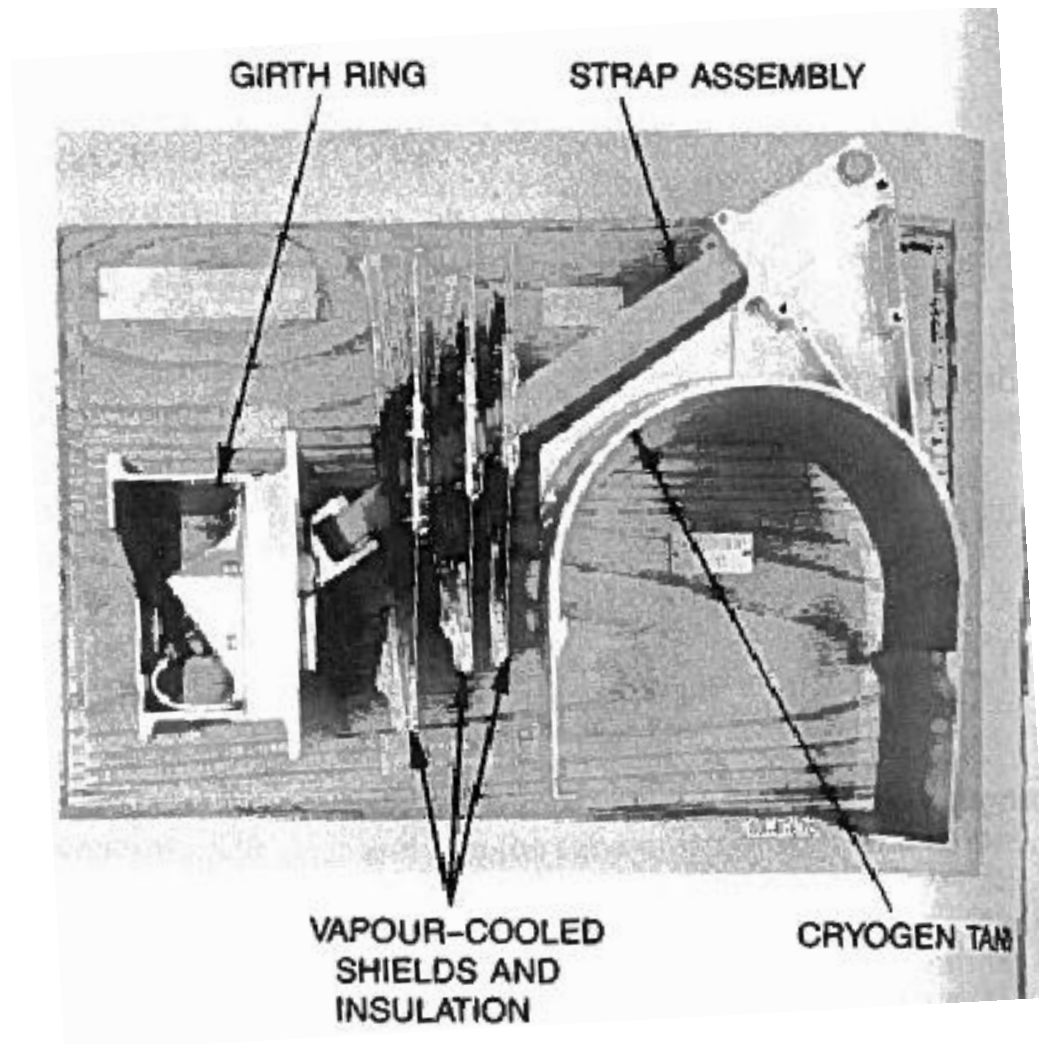


Figure 2 SHOOT support strap with friction clamp

K.F. Weintz et al. “SHOOT Dewar Support Strap Design”



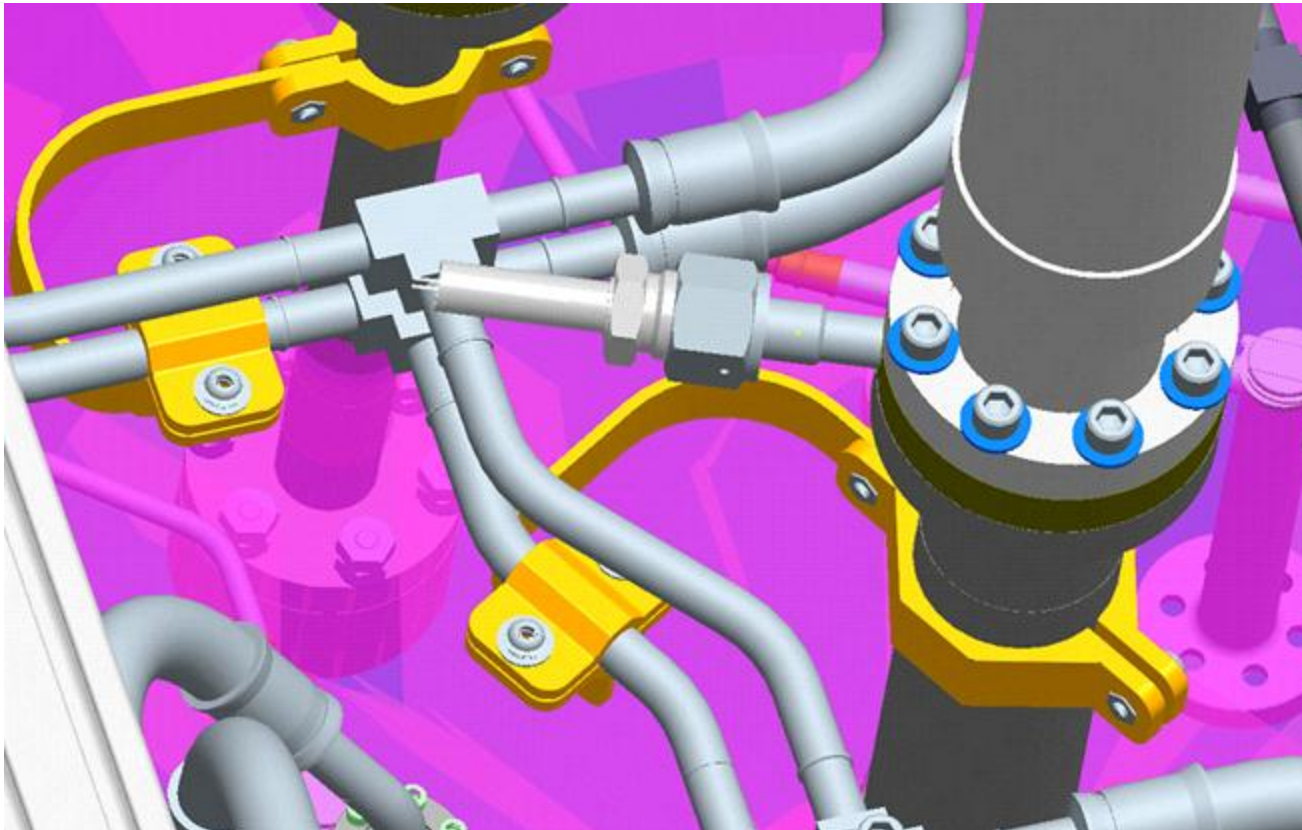
Cooling scheme application: shield mounted



T. Hirokawa et al., “Design of support strap with advanced composite for cryogenic application”

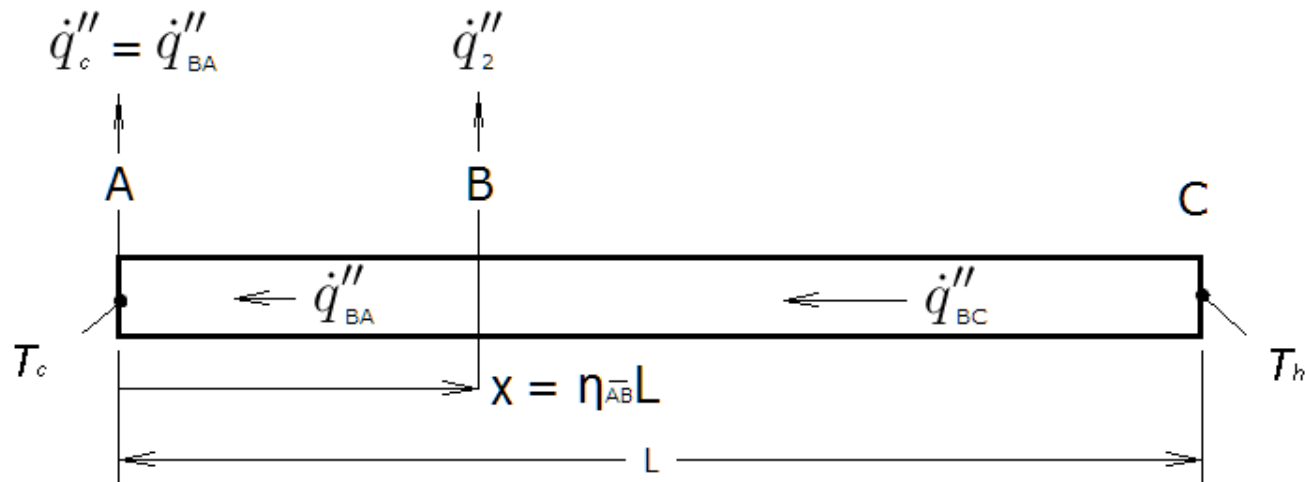
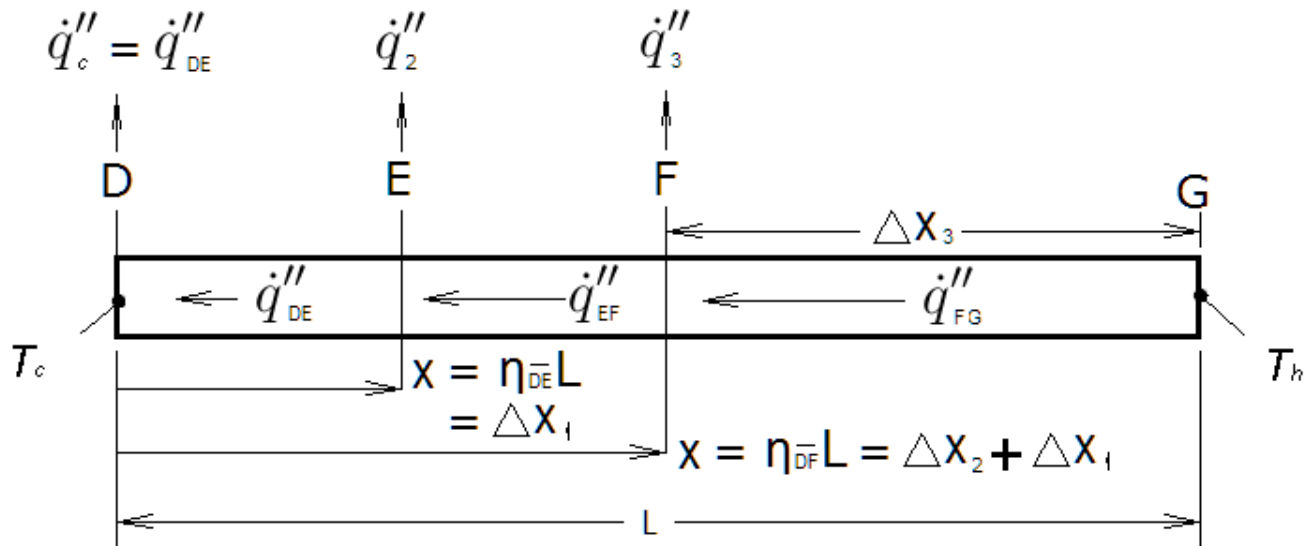
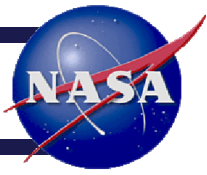


Another cooling scheme: thermal links



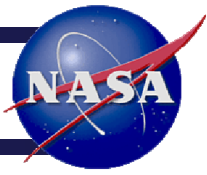


Discrete, two stage, perfect heat transfer





Hilal and Boom (1977)



230

M. A. Hilal and R. W. Boom

Table II. Optimum Temperatures, Locations, and Refrigeration Power for Finite Number of Shields

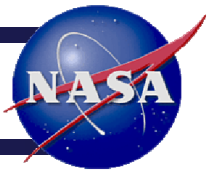
Number of shields	Cycle*	T_1, K	T_2, K	T_3, K	$\Delta x_1/L$	$\Delta x_2/L$	$\Delta x_3/L$	$\Delta x_4/L$	$PL/A,$ W/cm
304 Stainless steel ($T_c = 4.2 K$)									
1	C	39.7	—	—	0.338	0.662	—	—	445
1	A	39.7	—	—	0.338	0.662	—	—	1781
2	C	21.6	81.7	—	0.189	0.334	0.477	—	316

* C, Carnot cycle efficiencies; A, actual cycle efficiencies.

In my estimate, the second row is actually correct. Because solving the Carnot case gives $T_2 = 40K$ and $\eta_{\overline{AB}} = 0.351$ and when I tested a sample real coefficient of performance, I found roughly $T_2 = 39K$ and $\eta_{\overline{AB}} = 0.33$. So the first row—the Carnot case—was misprinted. Hilal and Boom actually had solutions up to four stages and also studied the same solutions for Narmco 570 cloth, but that information, being irrelevant, was omitted.



Hilal and Boom (1977)



230

M. A. Hilal and R. W. Boom

Table II. Optimum Temperatures, Locations, and Refrigeration Power for Finite Number of Shields

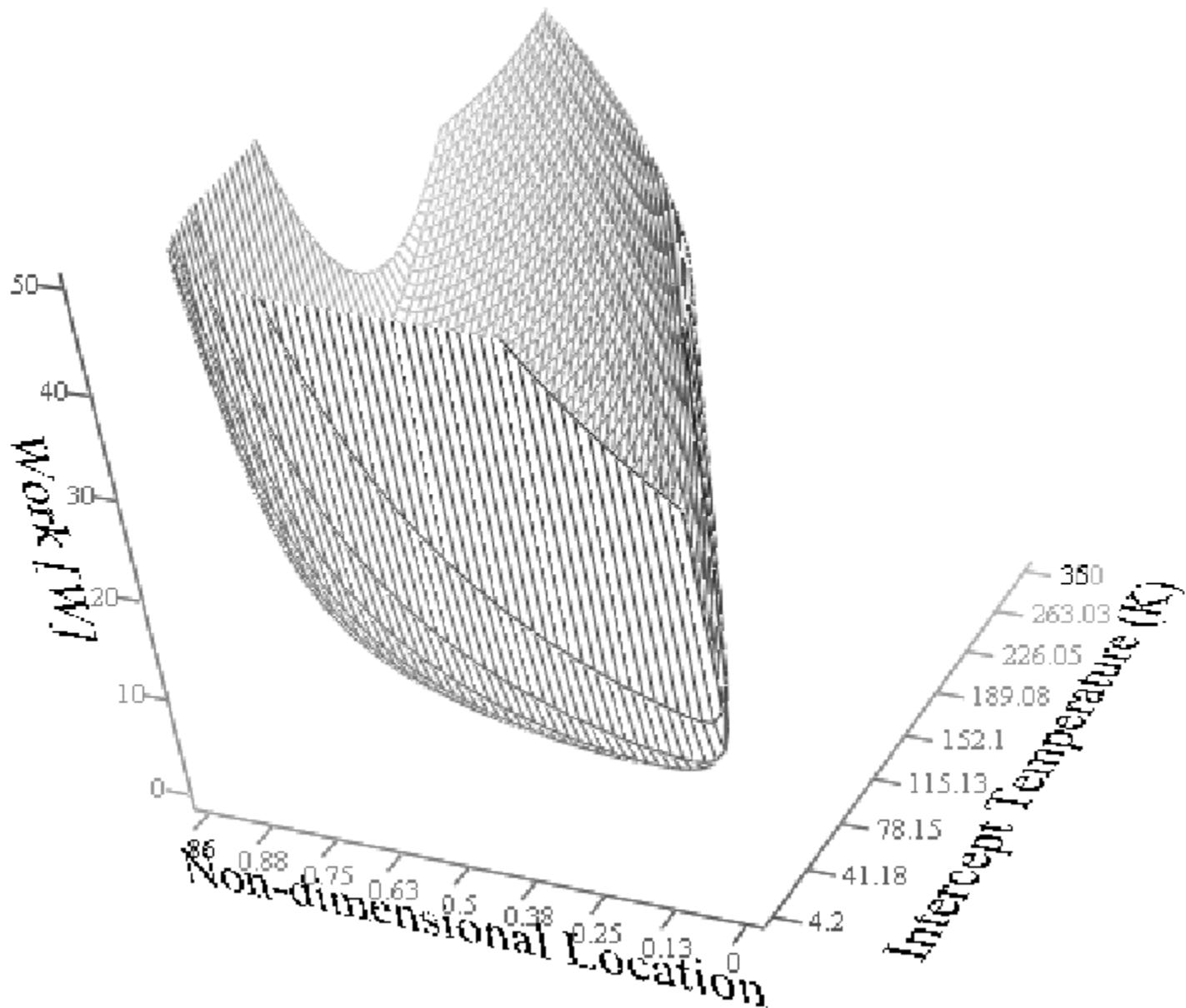
Number of shields	Cycle*	T_1, K	T_2, K	T_3, K	$\Delta x_1/L$	$\Delta x_2/L$	$\Delta x_3/L$	$\Delta x_4/L$	$PL/A, W/cm$
304 Stainless steel ($T_c = 4.2 K$)									
1	C	39.7	—	—	0.338	0.662	—	—	445
1	A	39.7	—	—	0.338	0.662	—	—	1781
2	C	21.6	81.7	—	0.189	0.334	0.477	—	316

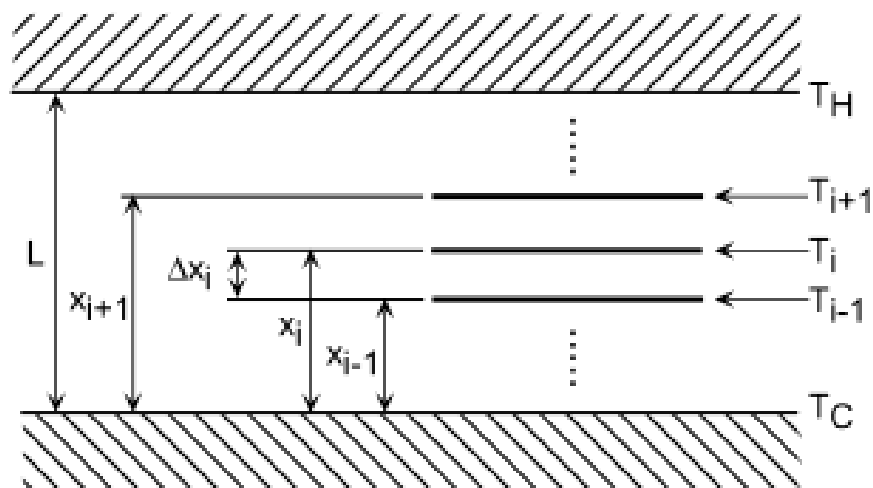
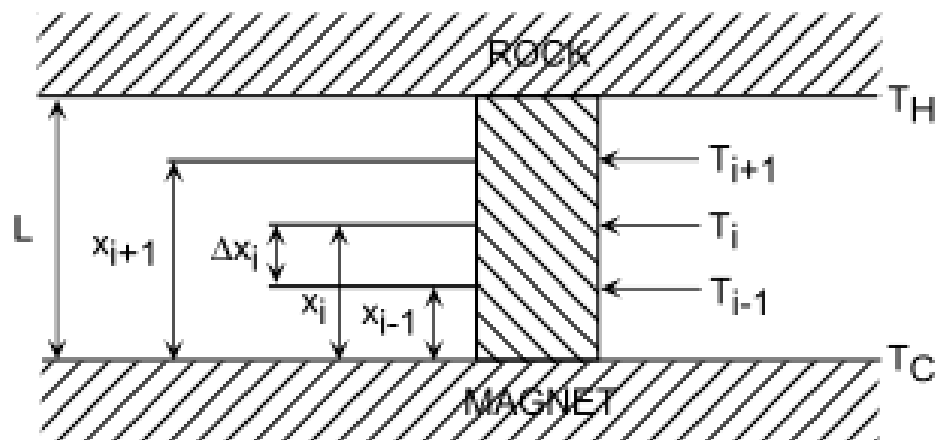
* C, Carnot cycle efficiencies; A, actual cycle efficiencies.

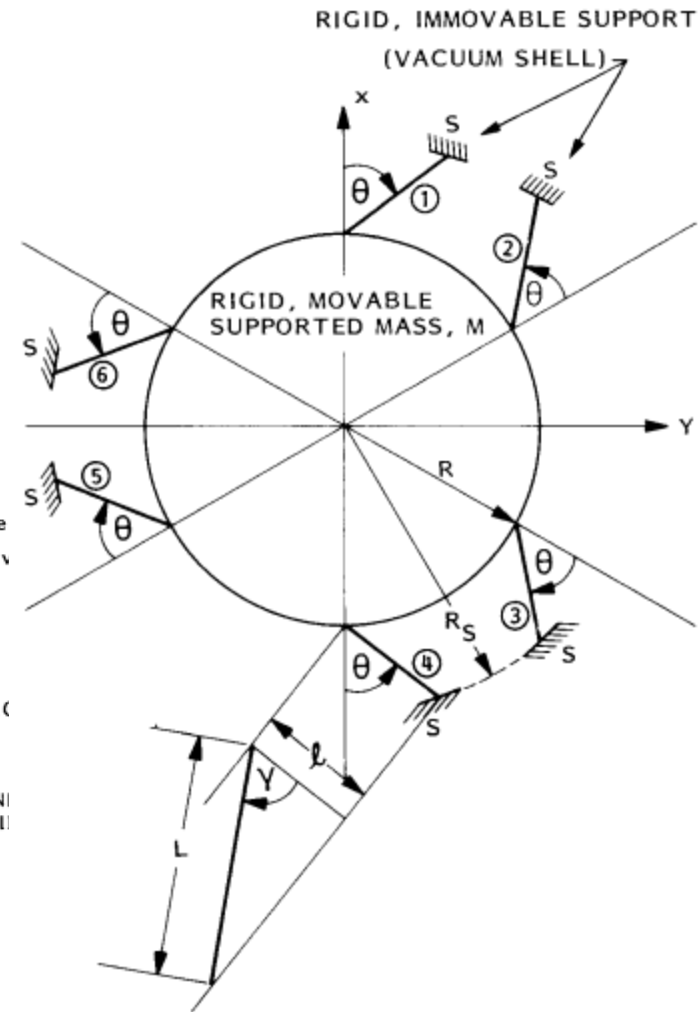
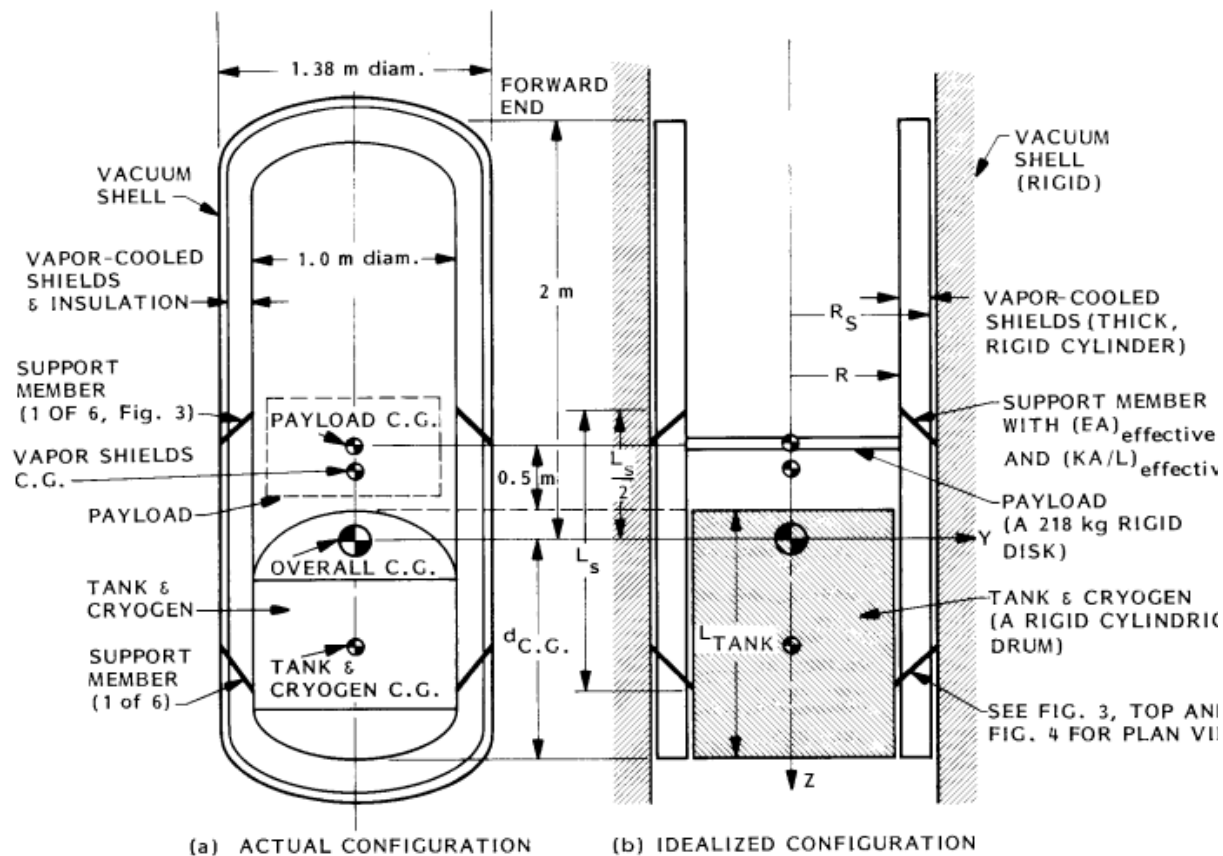
Number of Stages	$T_c (K)$	$T_2 (K)$	$T_3 (K)$	η_{AB}	η_{DE}	η_{DF}
2	4.2	40.0	-	0.351	-	-
3	4.2	21.1	80.7	-	0.194	0.524

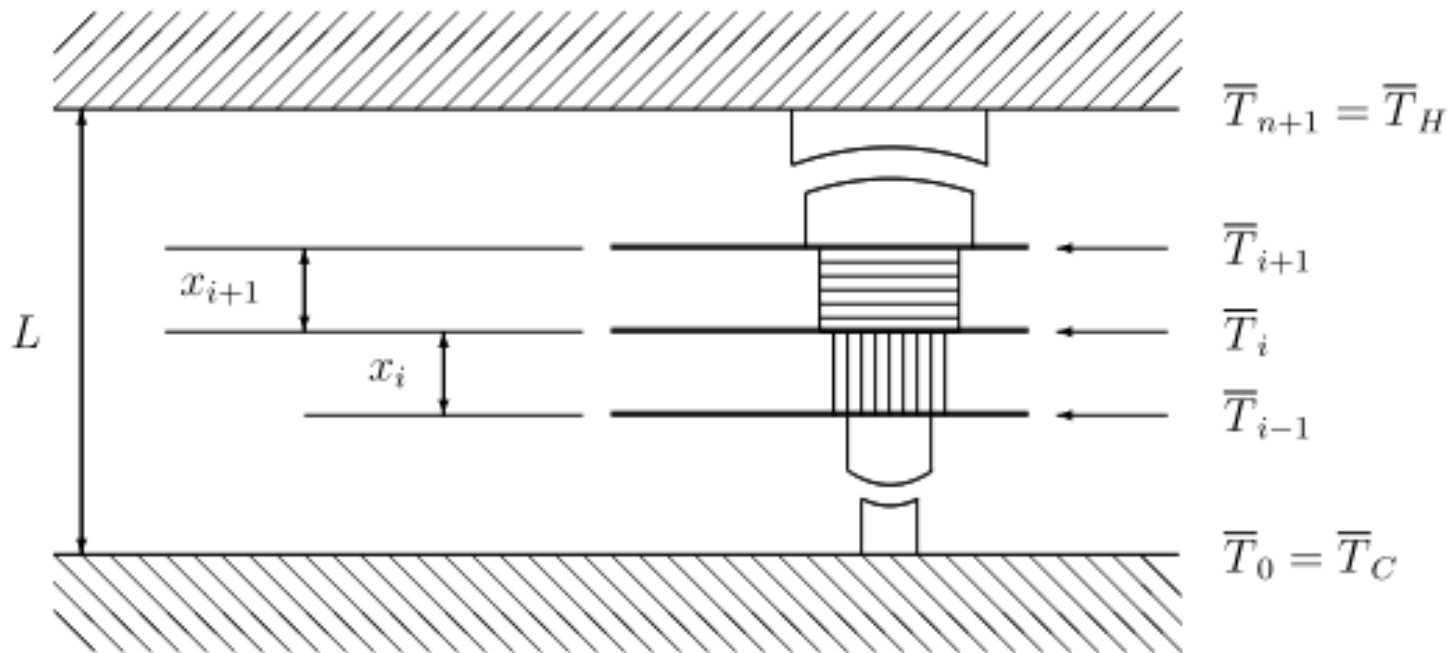


Carnot, two stage, ideal; graphical solution





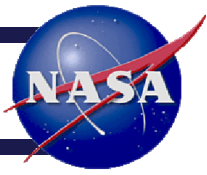








Bibliography



- All copyrighted works were adapted partially—never in whole—for fair use and informational purposes only. All pictures belong to the original owners and were shown in a private setting meant to spur thought, learning, and discussion. We do not use those works for commercial purposes nor do we claim credit for anything not expressly declared to have been created by the fine folks at the NASA Glenn Research Center. Anyone interested in more information about those resources will be able to find the original copyrighted works in the bibliography shown below.
- All opinions are those of the presenter and not necessarily those of NASA
- Kingsley, Wilson, Kirtley, Keim, Smith, Thullen, “Steady-state electrical tests on the MIT-ERPI 3-MVA Superconducting Generator”, IEEE Transactions on Power Apparatus and Systems, Vol. PAS-95, no. 3, May/June 1976

References

- [1] Mark A. Abramson. Mixed variable optimization of a load-bearing thermal insulation system using a filter pattern search algorithm. *Optim. Eng.*, 5 (2):157–177, 2002.
- [2] Paulo A. Augusto, Teresa Castelo-Grande, Pedro Augusto, and Domingos Barbosa. Optimization of refrigerated shields using multilayer thermal insulation: Cryostats design - analytical solution. *Cryogenics*, Jun 2006.
- [3] Adrian Bejan. *Improved thermal design of the cryogenic cooling system for a superconducting synchronous generator*. PhD thesis, Massachusetts Institute of Technology, Dec. 1974, handwritten edit ‘i.e. Feb. 1975’.
- [4] Adrian Bejan and J.L. Smith Jr. Thermodynamic optimisation of mechanical supports for cryogenic apparatus. *Cryogenics*, pages 158–163, 1974.
- [5] Adrian Bejan and Sylvie Lorente. The constructal law of design and evolution in nature. *Philosophical Transactions of The Royal Society*, pages 1335–1347, 2010.
- [6] Robert J. Boyle and Richard H. Knoll. Thermal analysis of shadow shields and structural members in a vacuum. Technical Note D-4876, NASA Lewis Research Center, Nov 1968.
- [7] David Bushnell. Optimum design of dewar supports. *J. Spacecraft*, 22(4):432–441, 1985.
- [8] E. R. Canavan and F. K. Miller. Optimized heat interception for cryogenic tank support. In *Advances in Cryogenic Engineering*, volume 53A and 53B, 2007.
- [9] J.C. Chato and J.M. Khodadidi. Optimization of cooled shields in insulations. *ASME Transactions, Journal of Heat Transfer*, 106 (4):871–875, 1984.
- [10] Cullimore and Ring, editors. *Sinaps User’s Manual*. Cullimore and Ring Technologies, Inc., 2011.
- [11] A.G. Fox and R.G. Scurlock. A letter to the editor. *Cryogenics*, Feb 1967. Department of Physics, University of Southampton, U.K.
- [12] M.A. Hilal and R.W. Boom. Optimization of mechanical supports for large superconductive magnets. In *Proceedings, International Cryogenic Materials Conference in Kingston, Ontario, Canada, 1975*, volume A77-42172 19-31, 1977.
- [13] M.A. Hilal and Y.M. Eyssa. Minimization of refrigeration power for large cryogenic systems. *Advances in Cryogenic Engineering*, 25, 1980.
- [14] Bergman Lavine Incropera, DeWitt. *Fundamentals of Heat and Mass Transfer*. Wiley, 6 edition, 2007.
- [15] Latif M. Jiji. *Heat Conduction*. Springer-Verlag Berlin Heidelberg, third edition, 2009. Founding Editor L.L. Faulkner.
- [16] Michael Kokkolaras, Charles Audet, and Jr. J.E. Dennis. Mixed variable optimization of the number and composition of heat intercepts in a thermal insulation system. *Engineering and Optimization*, 2 (1):5–29, 2001.
- [17] National Institute of Standards and Technology (NIST). Material properties. <http://cryogenics.nist.gov/MPropsMAY/materialproperties.htm>, 2012. Cryogenic Technologies Group.
- [18] A. R. Shouman. Nonlinear heat transfer and temperature distribution through fins and electric filaments of arbitrary geometry with temperature-dependent properties and heat generation. Technical report, NASA, Jan 1968.
- [19] Chi K. Tsao. Temperature distribution and power loss of a gas-cooled support for a cryogenic container. *Cryogenics*, 1974.